

CHAPTER 3

MASS CONSERVATION AND STREAM FUNCTIONS

3.1 CONTINUITY EQUATION

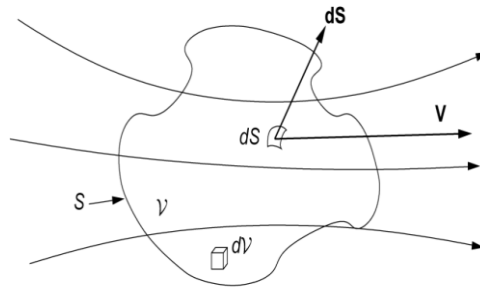


Fig 3.1 Finite control volume fixed in space

- Continuity equation is a statement of mass conservation converted into a differential equation
- From the Fig 3.1, rate of flow of mass into V , given as

$$-\int_S \rho \vec{v} \cdot d\vec{S}, \quad \rho = \rho(x, y, z, t)$$

where $\rho \vec{v}$ is a mass flux with density, ρ

- This integral measure the total fluid across the surface S per unit time
- Also rate of increases of mass into V , given as

$$\frac{d}{dt} \int_V \rho dV$$

And since V is a fix region this can be return as $\int_V \frac{d\rho}{dt} dV$

Due to mass is conserve, we must have

$$\int_V \frac{d\rho}{dt} dV = -\int_S \rho \vec{v} \cdot d\vec{S} \quad \dots\dots\dots(1)$$

Knowing the divergence theorem is

$$\int_V \nabla \cdot \vec{A} dV = \int_S \vec{A} \cdot d\vec{S}$$

so,

$$\int_s \rho \vec{v} \cdot dS = \int_V \nabla \cdot \rho \vec{v} dV \quad \dots\dots\dots(2)$$

Substitute (2) into (1)

$$\int_V \frac{d\rho}{dt} dV = - \int_V \nabla \cdot \rho \vec{v} dV$$

$$\int_V \left[\frac{d\rho}{dt} + \nabla \cdot \rho \vec{v} \right] dV = 0 \quad \dots\dots\dots(3)$$

Then from (3)

$$\frac{d\rho}{dt} + \nabla \cdot \rho \vec{v} = 0 \quad \dots\dots\dots(4)$$

Equation (4) is called continuity equation or mass conservation equation.

Notes:

1. If ρ is constant, then the continuity equation is $\nabla \cdot \vec{v} = 0$ (incompressible)
2. If ρ is independent of t , then the continuity equation is $\nabla \cdot \rho \vec{v} = 0$ (compressible)

3.2 THE CONVECTIVE DERIVATIVE

- In the case of incompressible fluid, ρ does not depend on \vec{v} , but each of the particles keep its density
- Suppose a particle of one position, \vec{r} and time t has density of $\rho(\vec{r}, t)$ when it moves and when $\vec{r} + \vec{v}\delta t$ and $t + \delta t$ the density must become $\rho(\vec{r} + \vec{v}\delta t, t + \delta t)$. So, clearly a change on density, ρ is

$$\rho(\vec{r} + \vec{v}\delta t, t + \delta t) - \rho(\vec{r}, t)$$

If we consider

$$\lim_{\delta t \rightarrow 0} \frac{\rho(\vec{r} + \vec{v}\delta t, t + \delta t) - \rho(\vec{r}, t)}{\delta t} = \vec{v} \cdot \nabla \rho + \frac{\partial \rho}{\partial t} \quad \text{(Taylor's theorem)}$$

Or we write

$$\frac{D\rho}{Dt} = \frac{\partial\rho}{\partial t} + \vec{v} \cdot \nabla\rho \dots\dots\dots(5)$$

Equation (5) is known as **convective derivative**

Using vector identity $\nabla \cdot \phi \vec{A} = \vec{A} \cdot \nabla \phi + \phi \nabla \cdot \vec{A}$ into (5), then

$$\frac{D\rho}{Dt} = \frac{\partial\rho}{\partial t} + \nabla \cdot \rho \vec{v} - \rho \nabla \cdot \vec{v}$$

or

$$\frac{D\rho}{Dt} + \rho \nabla \cdot \vec{v} = 0 \text{ since } \frac{\partial\rho}{\partial t} + \nabla \cdot \rho \vec{v} = 0$$

3.4 THE STREAM FUNCTION FOR 2D FLOW

- From the continuity equation for incompressible fluid $\nabla \cdot \vec{v} = 0$ means by using divergence theorem that is no total volume through and closed surface as much flows out as flow in.
- For the same of mathematics simply we choose to investigate the flow in which there are two non-zero velocity and two effective coordinates.
- There are:
 - i. Two-dimensional flow where the velocity defines as

$$\vec{v} = u(x, y)\hat{i} + v(x, y)\hat{j}$$

Or in polar plane

$$\vec{v} = V_r(r, \theta)\hat{r} + V_\theta(r, \theta)\hat{\theta}$$

- ii. Asymmetric flow (no swirl)

$$\vec{v} = V_r(r, z)\hat{r} + V_\theta(r, z)\hat{\theta}$$

Which shows that velocity not depend on θ

iii. In cylindrical coordinate, vector \vec{v} is independent of θ and there is no θ velocity. Such flow has two velocity components V_r, V_z and one differential equation $\nabla \cdot \vec{v} = 0$ connecting them

$$\vec{v} = V_r(r, z)\hat{r} + V_z(r, z)\hat{k}$$

4.3.1 Existence of Stream Function

- In 2D flow of incompressible fluid

$$\vec{v} = u(x, y)\hat{i} + v(x, y)\hat{j}$$

Continuity equation is given as

$$\nabla \cdot \vec{v} = 0 \quad \dots\dots\dots(9)$$

Equation (9) is fully satisfied equations

$$u = \frac{\partial \psi}{\partial y}, \quad v = -\frac{\partial \psi}{\partial x} \quad \dots\dots\dots(10),(11)$$

For any suitable differential function $\psi(x, y)$. Function $\psi(x, y)$ is known as **stream function**.

To proof:

$$\begin{aligned} \nabla \cdot \vec{v} &= \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} \\ &= \frac{\partial}{\partial x} \left(\frac{\partial \psi}{\partial y} \right) + \frac{\partial}{\partial y} \left(-\frac{\partial \psi}{\partial x} \right) \\ &= 0 \end{aligned}$$

To show that $\psi(x, y)$ exist and can be written in terms u and v , integrate $u = \frac{\partial \psi}{\partial y}$ with respect to y and let $y = N$ ($b \leq N \leq y$) where b is constant, gives

$$\psi = \int_b^y U(x, N) dN + \alpha(x) \quad \dots\dots\dots(12)$$

where $x(t)$ is a function of x and b is constant.

Differentiate (12) with respect to x

$$\frac{\partial \psi}{\partial x} = \int_b^y \frac{\partial}{\partial x} U(x, N) dN + \alpha'(x) \quad \dots\dots\dots(13)$$

From (9)

$$\begin{aligned} \nabla \cdot \vec{v} &= 0 \\ \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} &= 0 \\ \frac{\partial u}{\partial x} + \frac{\partial v}{\partial N} &= 0 \\ \frac{\partial u}{\partial x} &= -\frac{\partial v}{\partial N} \end{aligned}$$

so from (13)

$$\begin{aligned} \frac{\partial \psi}{\partial x} &= \int_b^y -\frac{\partial \psi}{\partial N} dN + \alpha'(x) \\ &= -v(x, y) + v(x, b) + \alpha'(x) \end{aligned}$$

Using (11)

$$\begin{aligned} -v(x, y) &= -v(x, y) + v(x, b) + \alpha'(x) \\ \alpha'(x) &= -v(x, b) \end{aligned}$$

Integrate $\alpha'(x)$ with respect to x and let $x = M$ ($a \leq M \leq x$)

$$\alpha(x) = -\int_a^M v(M, b) dM \quad \text{where } a \text{ is constant}$$

Then from (12),

$$\varphi = \int_b^y u(x, N) dN - \int_a^x v(M, b) dM . \text{ It shows that } \psi(x, y) \text{ exist in}$$

terms of u and v for any constants a and b .

➤ Determination of $u = \frac{\partial \psi}{\partial y}$ and $v = -\frac{\partial \psi}{\partial x}$

- If $\nabla \cdot \vec{B} = 0$, so there exists a vector \vec{A} such that $\vec{B} = \nabla \times \vec{A}$, \vec{A} is called the vector potential of \vec{B} .

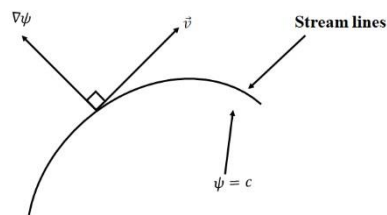
- By using definition, we can say that $\nabla \cdot \vec{v} = 0$, there exist \vec{A} such that $\vec{v} = \nabla \times \vec{A}$
- If we write $\vec{A} = \psi(x, y)\hat{k}$ then

$$\begin{aligned}\vec{v} &= \nabla \times \vec{A} \\ &= \\ &= \end{aligned}$$

$$\Rightarrow u = \frac{\partial \psi}{\partial y} \text{ and } v = -\frac{\partial \psi}{\partial x}$$

4.4 PROPERTIES OF STREAM FUNCTION

Let $\vec{v} = \nabla \psi + \hat{k}$



Vector, \vec{v} is perpendicular to $\nabla \psi$ and $\nabla \psi$ is perpendicular to $\psi = c$ where c is constant. Therefore, vector \vec{v} is parallel to the curve $\psi = c$.
Curve $\psi = c$ are streamlines.

Example 1:

Given $\psi(x, y) = Uy$, where U is a constant. Sketch the streamlines.

4.5 SOME BASIC STREAM FUNCTIONS

a) Flows parallel to the x-axis

i. *Uniform stream of speed U parallel to the x-axis*

Let $\vec{v} = U\hat{i} + 0\hat{j}$, U is constant. Find $\psi(x, y)$

Solution:

ii. *Uniform shear flow*

Let $\vec{v} = \beta y\hat{i} + 0\hat{j}$, β is constant. Find $\psi(x, y)$

iii. *First flow near a wall in channel*

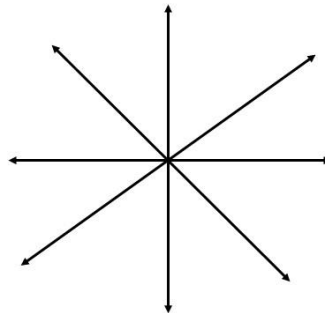
Let $\vec{v} = U \ln\left(\frac{y}{a}\right)\hat{i} + 0\hat{j}$ where U and a are constant. Find $\psi(x, y)$

iv. *Transition layer*

Let $\vec{v} = \left(U_0 + \frac{1}{2}(U_1 - U_0) \left[1 + \tanh\left(\frac{y}{a}\right) \right] \right) \hat{i} + 0\hat{j}$ where U_0 , U_1 and a are constant. Find $\psi(x, y)$.

b) Flow radially outwards

Flow leaves the vicinity/ source radially with the same speed at all angles .



Given $\vec{v} = f(r)\hat{r} + 0\hat{\theta}$. We need to consider stream function in plane polar coordinates.

In polar coordinates,

$$\vec{v} = V_r\hat{r} + V_\theta\hat{\theta}$$

To show the relationship between velocity, v and stream function .. in polar coordinates

$$\vec{v} = \nabla\psi \times \hat{k}$$

We need to use cylindrical polar coordinates,

Given $\vec{v} = f(r)\hat{r} + 0\hat{\theta}$. Find $\psi(x, y)$

$$\frac{\partial \psi}{\partial r} = 0 \quad \dots\dots\dots(1)$$

$$\frac{1}{r} \frac{\partial \psi}{\partial \theta} = f(r) \quad \dots\dots\dots(2)$$

We can show that

$$\psi = rf(r)\theta \quad \dots\dots\dots(3)$$

But to satisfy (1), $\psi(r, \theta)$ is independent of r and $\frac{\partial \psi}{\partial \theta}$ must be independent of θ .

Let $\frac{\partial \psi}{\partial \theta} = A$

Since

$$V_r = \frac{1}{r} \frac{\partial \psi}{\partial \theta}$$

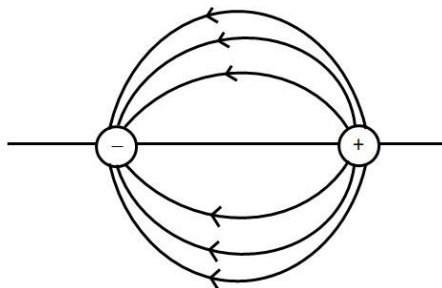
$$V_r = \frac{1}{r} A$$

$$\therefore f(r) = \frac{1}{r} A$$

From (3),

$$\begin{aligned} \psi &= r \left(\frac{1}{r} A \right) \theta \\ &= A\theta \end{aligned}$$

c) A dipole along the x-axis



Given velocity

$$\vec{v}(r, \theta) = 0\hat{r} + V_\theta(r, \theta)\hat{\theta}$$

The appropriate stream function is $\psi(r, \theta) = \frac{\mu}{r} \sin \theta$ where μ is a constant

d) Fluid in circle

Given velocity

$$\vec{v}(r, \theta) = 0\hat{r} + f(r)\hat{\theta}$$

There are two special cases of $f(r)$:

- i. $V_\theta = f(r) = \frac{c}{r}$ (line vortex)
- ii. $V_\theta = f(r) = Dr$ (rigid body motion)

We know that,

$$\begin{aligned} V_\theta &= -\frac{\partial \psi}{\partial r} \\ \vec{v} &= V_r\hat{r} + V_\theta\hat{\theta} \\ V_r &= \frac{1}{r} \frac{\partial \psi}{\partial \theta} \\ V_\theta &= -\frac{\partial \psi}{\partial r} \\ -\frac{\partial \psi}{\partial r} &= Dr \\ \psi &= -\frac{Dr^2}{2} + A(\theta) \end{aligned}$$

From (i),

$$\begin{aligned} -\frac{\partial \psi}{\partial r} &= \frac{c}{r} \\ \psi &= -c \ln\left(\frac{r}{a}\right) ; \quad c = \frac{k}{2\pi}, \text{ where } k \text{ is circulation} \end{aligned}$$

Therefore line vortex has stream function, $\psi = -\frac{k}{2\pi} \ln\left(\frac{r}{a}\right)$.

3.6 THE STREAM FUNCTION FOR AXISYMMETRIC FLOW (STOKES)

In axisymmetric flow,

$$\vec{v}(r, z) = U_r(r, z)\hat{r} + V_z(r, z)\hat{k}$$

a) *Existence of Stokes stream function, $\psi(r, z)$*

Hence, we used cylindrical coordinates by taking

$$\vec{v}(r, z) = U_r(r, z)\hat{r} + V_z(r, z)\hat{k}$$

And using continuity equation, $\nabla \cdot \vec{v} = 0$

We may consider, $\frac{\partial}{\partial r}(rU_r) + \frac{\partial}{\partial z}(rV_z) = 0$

Therefore, it can be shown that the stream function in terms of U_r and V_z is

b) *Determination of axisymmetric stream function in cylindrical and spherical coordinate*

Since $\nabla \cdot \vec{v} = 0$, then axis vector \vec{A} such that $\vec{v} = \nabla \times \vec{A}$ where \vec{A} is vector potential.

Let $\vec{A} = \frac{1}{r}\psi\theta$.

Therefore, by using cylindrical coordinates,

In axisymmetric flow, $\vec{v} = V_r(r, z)\hat{r} + V_z(r, z)\hat{k}$

$$\therefore V_r = -\frac{1}{r} \frac{\partial \psi}{\partial z}$$

$$\therefore V_z = \frac{1}{r} \frac{\partial \psi}{\partial r}$$

Also, since $\nabla \cdot \vec{v} = 0$, then there exist \vec{A} such that $\vec{v} = \nabla \times \vec{A}$ where \vec{A} is vector potential

$$\vec{A} = \frac{1}{r \sin \theta} \psi \hat{\lambda}$$

Therefore, in spherical coordinates,

In spherical we have,

$$\therefore V_r = \frac{1}{r^2 \sin \theta} \frac{\partial \psi}{\partial \theta}$$

$$\therefore V_z = -\frac{1}{r \sin \theta} \frac{\partial \psi}{\partial r}$$