

CHAPTER 6 (PART II)

SOLUTION OF NAVIER STOKES EQUATION

6.1 EXACT SOLUTION OF NAVIER STOKES EQUATION

c) Unsteady Flow due to Oscillating Flow

Consider also two-dimensional flow moving in x -direction (unidirectional flow)

The general solution is obtained similar to problem (b) as

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2} \text{ ----- (i)}$$

but the boundary conditions become

$$u(0,t) = u \cos \omega t \text{ ----- (ii)}$$

$$u(\infty,t) \text{ is bounded ----- (iii)}$$

To solve, need to consider method of separation of variables by taking

$$u(y,t) = e^{i\omega t} f(y)$$

and substitute into (i)

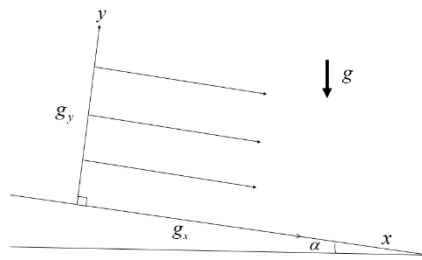
Second order homogeneous equation. Take $f = e^{m\eta}$

$$u(y,t) = e^{i\omega t} \left[Ae^{\sqrt{\frac{\omega}{2\nu}}(1+i)y} + Be^{-\sqrt{\frac{\omega}{2\nu}}(1+i)y} \right] \text{----- (iv)}$$

After using (2) and (3), the solution is
 u is bounded if $y \rightarrow \infty$.

$$u = u \cos \omega t , \quad y = 0 .$$

d) Steady Flow under Gravity Down on Inclined Plane



$$\sin \alpha = \frac{g_x}{g} \quad \cos \alpha = \frac{g_y}{g}$$

Consider the fluid as two-dimensional incompressible uni-directional flow.

Simplification:

1. Steady - not depend on t
2. 2D - depend on x and y
 - $u(x, y)$ and $v(x, y)$
3. Uni-directional flow $v(x, y) = 0$

From continuity equation:

$$\Rightarrow u = u(y)$$

From Equation (2) (Chapter 6 Part I)

$$0 = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \frac{\partial^2 u}{\partial y^2} + g_x \text{ ----- (i)}$$

From Equation (3) (Chapter 6 Part I)

$$0 = -\frac{1}{\rho} \frac{\partial p}{\partial y} + g_y \text{ ----- (ii)}$$

Therefore, since $g_y = g \cos \alpha$

$$\frac{\partial p}{\partial y} = \rho g \cos \alpha$$

$$p(x, y) = y \rho g \cos \alpha + f(x) \text{ ----- (iii)}$$

Equation (4) (Chapter 6 Part I) fully eliminated.

Note:

From (iii)

From (i)

Therefore,

Boundary conditions

$$u = 0 \quad \text{at } y = 0$$
$$\frac{\partial u}{\partial y} = 0 \quad \text{at } y = h$$

Find A and B using BCs,

Then the solution becomes

e) Flow in a Pipe

Navier-Stokes equation in cylindrical coordinate is given as continuity equation

$$\frac{1}{r} \frac{\partial}{\partial r}(rU_r) + \frac{1}{r} \frac{\partial U_\theta}{\partial \theta} + \frac{\partial U_z}{\partial z} = 0 \text{ ----- (i)}$$

where $\vec{v} = (U_r, U_\theta, U_z)$

Momentum equation:

x -component

$$U_r \frac{\partial U_r}{\partial r} + \frac{U_\theta}{r} \frac{\partial U_r}{\partial \theta} - \frac{U_\theta^2}{r} + U_z \frac{\partial U_r}{\partial z} = -\frac{1}{\rho} \frac{\partial p}{\partial r} + \frac{\mu}{\rho} \left(\nabla^2 U_r - \frac{U_r}{r^2} - \frac{2}{r^2} \frac{\partial U_\theta}{\partial \theta} \right) \text{ ----- (ii)}$$

y -component

$$U_r \frac{\partial U_\theta}{\partial r} + \frac{U_\theta}{r} \frac{\partial U_\theta}{\partial \theta} - \frac{U_r U_\theta}{r} + U_z \frac{\partial U_\theta}{\partial z} = -\frac{1}{\rho} \frac{\partial p}{\partial \theta} + \frac{\mu}{\rho} \left(\nabla^2 U_\theta - \frac{U_\theta}{r^2} + \frac{2}{r^2} \frac{\partial U_r}{\partial \theta} \right) \text{ ----- (iii)}$$

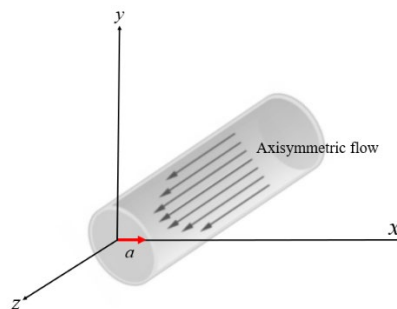
z -component

$$U_r \frac{\partial U_z}{\partial r} + \frac{U_\theta}{r} \frac{\partial U_z}{\partial \theta} + U_z \frac{\partial U_z}{\partial z} = -\frac{1}{\rho} \frac{\partial p}{\partial z} + \frac{\mu}{\rho} (\nabla^2 U_z) \text{ ----- (iv)}$$

$$\text{where } \nabla^2 = \frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial}{\partial r} \right) + \frac{1}{r^2} \frac{\partial^2}{\partial \theta^2} + \frac{\partial^2}{\partial z^2}$$

Case I: Steady flow in a pipe

- Considered a fully developed laminar motion through a tube of radius, a
- Flow through a tube is frequently called circular Poissule flow
- To solve the problem cylindrical coordinate (r, θ, z) need to be applied with the z axis coinciding with the axis of the pipe



- In this flow problem only velocity $U_z(r, \theta, z)$ is known zero components of velocity as the other $U_r = U_\theta = 0$

Continuity equation:

$$\frac{1}{r} \frac{\partial}{\partial r} (r U_r) + \frac{1}{r} \frac{\partial U_\theta}{\partial \theta} + \frac{\partial U_z}{\partial z} = 0$$

$$\frac{\partial U_z}{\partial z} = 0$$

$$U_z = U_z(r, \theta)$$

But axisymmetric flow (not depend on θ), then

$$U_z = U_z(r)$$

Momentum equation:

x -component

y -component

z -component

The equation of motion becomes:

$$\frac{\partial p}{\partial z} = \mu \left(\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial U_z}{\partial r} \right) \right) \text{----- (v)}$$

And the boundary condition

U_z is bounded at $r = 0$

$U_z = 0$ at $r = a$

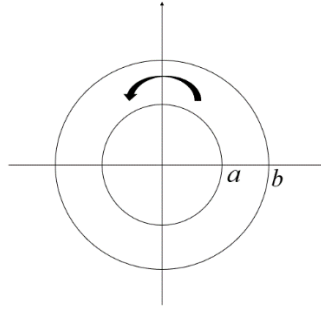
Solution of (v):

And boundary conditions:

U_z is bounded at $r = 0$

$U_z = 0$ at $r = a$

Case II: Steady flow between two rotating cylinders



- Consider the steady 2D velocity $\vec{U} = U_\theta \hat{\theta}$ in between concentric cylinder with radius $r = a$ and $r = b$ respectively and $a < b$
- In this concentric cylinder, the flow can only be steady if at least one of the cylinders is forced to rotate
- Therefore, the boundary condition can be stated as
 - a) By considering continuity equation in cylindrical coordinate, show that

$$U_\theta = U_\theta(r, z)$$

- b) By considering momentum equation in cylindrical coordinate, show that the equation of motion is

$$-\frac{U_\theta^2}{2} = \frac{1}{\rho} \frac{\partial p}{\partial r}$$
$$0 = \frac{\mu}{\rho} \left[\frac{1}{r} \frac{\partial}{\partial r} \left(r \frac{\partial U_\theta}{\partial r} \right) - \frac{U_\theta}{r^2} \right]$$

- c) Show that by solving equation in

$$U_\theta(r) = -\frac{\Omega a^2}{b^2 - a^2} r + \frac{\Omega a^2 b^2}{(b^2 - a^2) r}$$