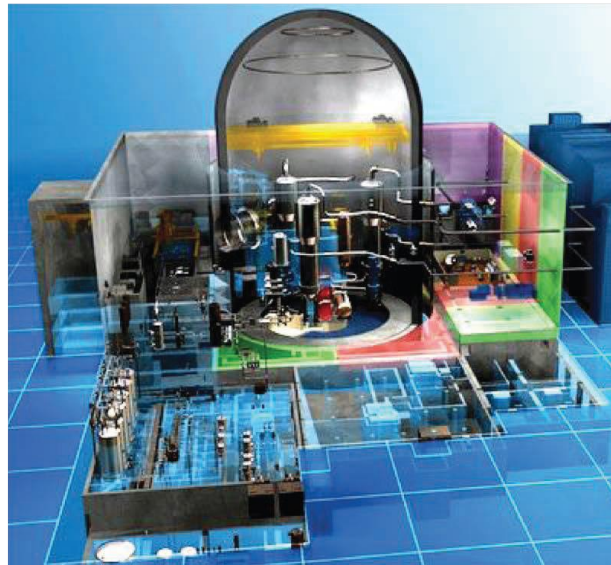




Heat Conduction in Reactor Elements (Part 4)



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Heat Flow Out of Solid Cylindrical Fuel Elements

- **Unclad cylindrical fuel element**, or **fuel rod** or **pin**, of a **diameter small** compared with that of the reactor core will be treated.
- The **neutron flux** will be assumed **not to change** appreciably in **either** the **axial** or **radial** directions.
- The **heat flow out** of the fuel element will **substantially** be **radial** and will be **equal** in all **directions**.
- The **equation** here is the one-dimensional **Poisson equation** in cylindrical coordinates, given by

$$\nabla^2 T + \frac{q'''}{k} = 0 \quad \text{where} \quad \nabla^2 T = \left(\frac{d^2 T}{dr^2} + \frac{1}{r} \frac{dT}{dr} \right)$$



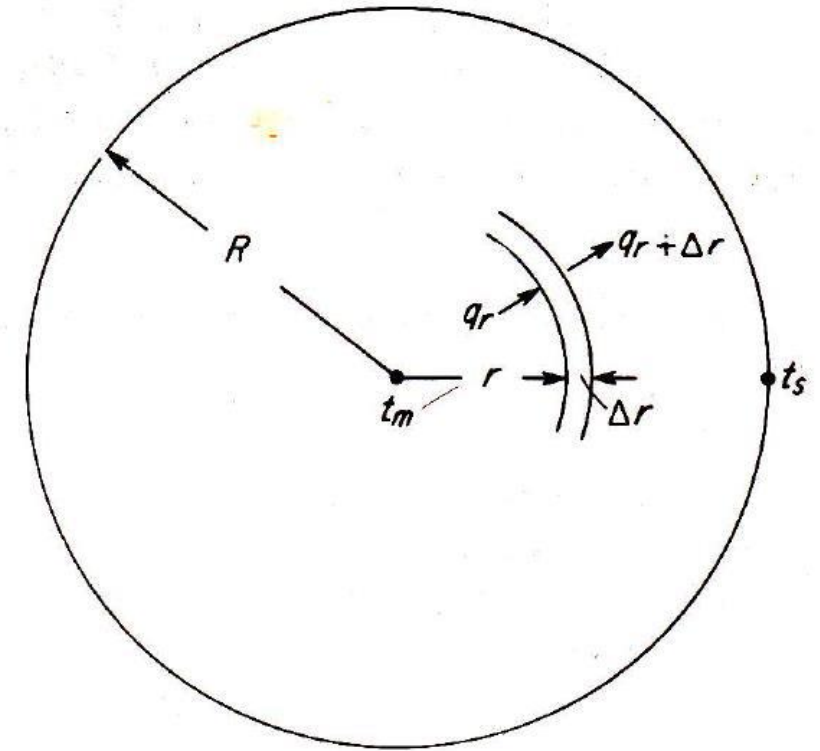
Heat Flow Out of Solid Cylindrical Fuel Elements

$$\nabla^2 T + \frac{q'''}{k} = 0$$

$$\nabla^2 T = \left(\frac{d^2 T}{dr^2} + \frac{1}{r} \frac{dT}{dr} \right)$$

- This **equation** can also be obtained by a **heat balance** on a cylindrical shell within the **fuel element**.
- The cross section of an **unclad element** having a **radius R** and **length L**
- A **thin cylindrical** shell at **r** of thickness **Δr** .
- A **heat balance** of **steady-state** heat transfer as follows:

$$q'''(2\pi r \Delta r L) = q_{r+\Delta r} - q_r$$



Heat Flow Out of Solid Cylindrical Fuel Elements

$$q'''(2\pi r \Delta r L) = q_{r+\Delta r} - q_r \quad (1)$$

where

$$q_r = -Ak_f \frac{dT}{dr} = -2\pi r L k_f \frac{dT}{dr} \quad (2)$$

and

$$q_{r+\Delta r} = q_r + \frac{dq_r}{dr} \Delta r$$

$$q_{r+\Delta r} = q_r - 2\pi L k_f \Delta r \frac{d}{dr} \left(r \frac{dT}{dr} \right)$$

$$q_{r+\Delta r} = q_r - 2\pi L k_f \left(r \frac{d^2 T}{dr^2} + \frac{dT}{dr} \right) \Delta r$$

$$q_{r+\Delta r} - q_r = -2\pi L k_f \left(r \frac{d^2 T}{dr^2} + \frac{dT}{dr} \right) \Delta r \quad (3)$$



Heat Flow Out of Solid Cylindrical Fuel Elements

Substitute equation (2) and (3) in (1)

$$q'''(2\pi r \Delta r L) = -2\pi L k_f \left(r \frac{d^2 T}{dr^2} + \frac{dT}{dr} \right) \Delta r$$

$$q''' r = -k_f \left(r \frac{d^2 T}{dr^2} + \frac{dT}{dr} \right)$$

$$q''' r = -k_f \frac{d}{dr} \left(r \frac{dT}{dr} \right) \quad (4)$$



Heat Flow Out of Solid Cylindrical Fuel Elements

$$\nabla^2 T + \frac{q'''}{k} = 0 \qquad \nabla^2 T = \left(\frac{d^2 T}{dr^2} + \frac{1}{r} \frac{dT}{dr} \right)$$

$$q'''' r = -k_f \left(r \frac{d^2 T}{dr^2} + \frac{dT}{dr} \right)$$

$$\frac{q''''}{k_f} = -\frac{1}{r} \left(r \frac{d^2 T}{dr^2} + \frac{dT}{dr} \right)$$

$$\frac{q''''}{k_f} = - \left(\frac{d^2 T}{dr^2} + \frac{1}{r} \frac{dT}{dr} \right)$$

This reduces to

$$\frac{d^2 T}{dr^2} + \frac{1}{r} \frac{dT}{dr} + \frac{q''''}{k_f} = 0$$



Heat Flow Out of Solid Cylindrical Fuel Elements

$$q'''r = -k_f \frac{d}{dr} \left(r \frac{dT}{dr} \right)$$

$$\frac{d^2T}{dr^2} + \frac{1}{r} \frac{dT}{dr} + \frac{q'''}{k_f} = 0$$

$$\frac{d}{dr} \left(r \frac{dT}{dr} \right) + \frac{q'''}{k_f} r = 0$$

$$\frac{d}{dr} \left(r \frac{dT}{dr} \right) = -\frac{q'''}{k_f} r$$

Integrate 2 times

$$\left(r \frac{dT}{dr} \right) = -\frac{q'''}{2k_f} r^2 + C_1$$

$$\frac{dT}{dr} = -\frac{q'''}{2k_f} r + \frac{C_1}{r}$$



Heat Flow Out of Solid Cylindrical Fuel Elements

$$\frac{dT}{dr} = -\frac{q'''}{2k_f}r + \frac{C_1}{r}$$

$$T(r) = -q''' \frac{r^2}{4k_f} + C_1 \ln r + C_2 \quad (5)$$

C_1 and C_2 , the constants of the double integration. Hence, apply boundary condition as follows:

Boundary condition 1

$$\frac{dT}{dr} = 0 \quad \text{at } r = 0 \quad \quad 0 = -\frac{q'''}{2k_f}(0) + \frac{C_1}{r} \quad \longrightarrow \quad C_1 = 0$$

Boundary condition 2

$$T = T_m \quad \text{at } r = 0 \quad \quad T_m = T(0) = -q''' \frac{(0)^2}{4k_f} + (0) \ln (0) + C_2 \quad \longrightarrow \quad C_2 = T_m$$



Heat Flow Out of Solid Cylindrical Fuel Elements

Hence,

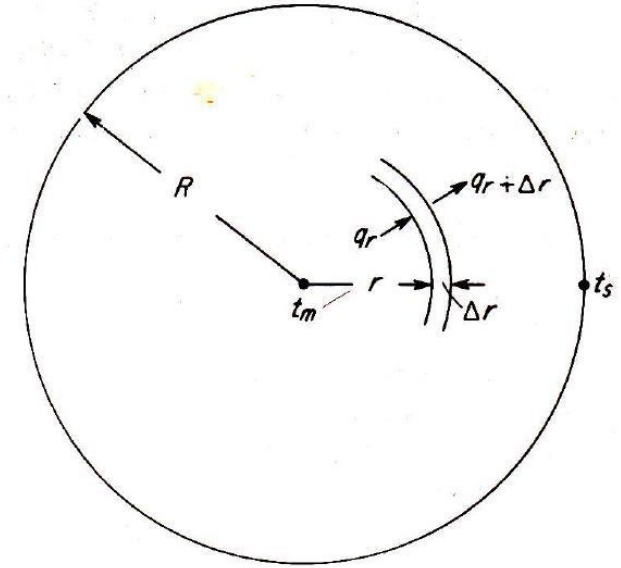
$$T(r) = -q''' \frac{r^2}{4k_f} + T_m$$

Solution expressing T in terms of r is,

$$T(r) = T_m - \frac{q''' r^2}{4k_f} \quad (6)$$

The maximum temperature difference in the system may be obtained by putting $r = R$ and $T = T_s$

$$T_m - T_s = \frac{q''' R^2}{4k_f} \quad (7)$$





Heat Flow Out of Solid Cylindrical Fuel Elements

The heat flow $q(r)$ at any radius r is equal to the total heat generated by the element within that radius. Thus

$$q(r) = \pi r^2 L q''' \quad (8)$$

$$T_m - T_s = \frac{q''' R^2}{4k_f}$$

The heat conducted out of the periphery of the element q_s , is equal to the heat generated in the entire element
And is given by:

$$q_s = \pi R^2 L q''' \quad q''' = \frac{q_s}{\pi R^2 L} \quad (9)$$

$$T_m - T_s = \frac{R^2}{4k_f} \times \frac{q_s}{\pi R^2 L} = \frac{1}{4k_f} \frac{q_s}{\pi L} \quad q_s = 4\pi L k_f (T_m - T_s) \quad (10)$$



Heat Flow Out of Solid Cylindrical Fuel Elements

Total heat generated is only a function of the maximum temperature drop ($T_m - T_s$) and is independent of the volumetric

Putting $2\pi RL = A_s$, the total heat generated

$$A_s = 2\pi RL \quad \frac{A_s}{R} = 2\pi L \quad \frac{2A_s}{R} = 4\pi L$$

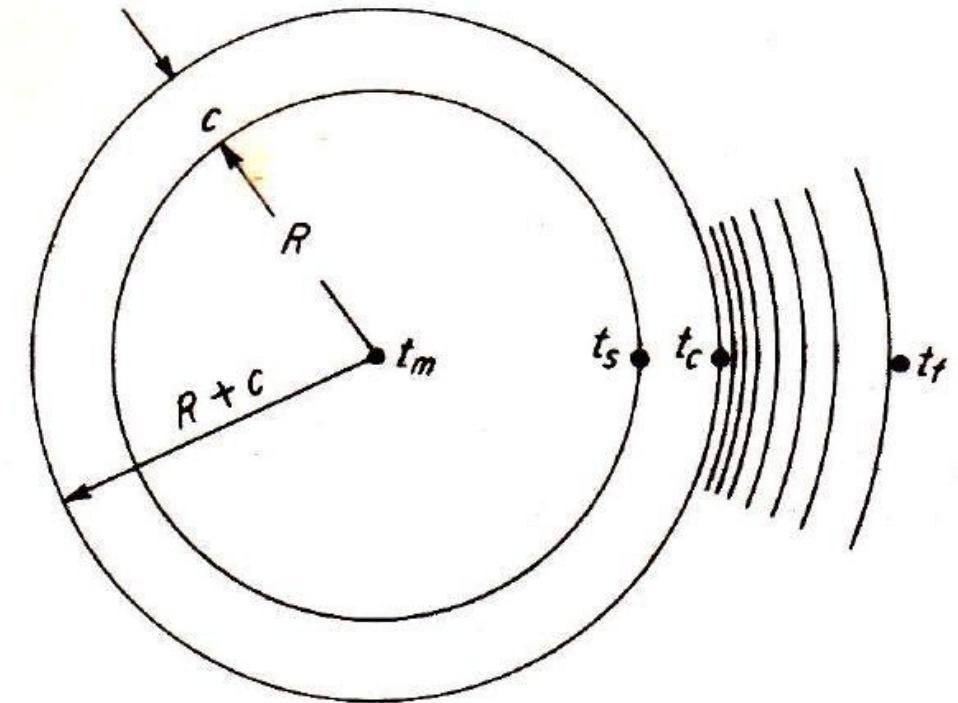
$$q_s = 4\pi L k_f (T_m - T_s) \quad q_s = \frac{2A_s}{R} k_f (T_m - T_s)$$

$$q_s = 2k_f A_s \left(\frac{T_m - T_s}{R} \right) \quad (11)$$



Effect of Cladding and Coolant

- The cross section of a **cylindrical fuel element** of radius R and axial length L , having a center temperature T_m and surrounded by cladding of radial thickness c and coolant fluid having bulk temperature T_f
- In the **steady state** with **no heat generated** in **cladding** or **coolant**,
- The heat **passing out** of the **fuel surface** is the same as that **passing through** the **cladding** and **into** the **coolant** fluid





Effect of Cladding and Coolant

The heat conduction equation for the cladding is

$$\frac{d^2T}{dr^2} + \frac{1}{r} \frac{dT}{dr} = 0 \quad (12)$$

which has the solution

$$T(r) = C_1 \ln(r) + C_2 \quad (13)$$

Using the boundary conditions

B. C (1)

$$T(R) = T_s$$

B. C (2)

$$T(R + c) = T_c$$

to determine the constants C_1 and C_2 gives

$$T(r) = \frac{T_s \ln(R + C) - T_c \ln(R) - (T_s - T_c) \ln(r)}{\ln\left(1 + \frac{C}{R}\right)} \quad (14)$$



Effect of Cladding and Coolant

$$T(r) = \frac{T_s \ln(R + C) - T_c \ln(R) - (T_s - T_c) \ln(r)}{\ln\left(1 + \frac{C}{R}\right)}$$

From Fourier's law, the total heat flowing out of the cladding is

$$q_s = -k_c A \frac{dT}{dr}$$

$$A_{R+c} = 2\pi L(R + c) \quad (15)$$

$$q_s = -2\pi r L k_c \frac{dT}{dr} \quad (16)$$

evaluated at $r = R + c$. Carrying out the differentiation of equation (14) yields

$$q_s = 2\pi L k_c \frac{(T_s - T_c)}{\ln\left(\frac{R + c}{R}\right)} \quad (17)$$



Effect of Cladding and Coolant

The expression for heat flow through the cladding is that for heat flow by conduction through a hollow cylinder.

$$q_s = -4\pi Lk_f(T_m - T_s)$$

$$A_R = 2\pi LR$$

For cladding

$$q_s = 2\pi Lk_c \frac{(T_s - T_c)}{\ln\left(\frac{(R + c)}{R}\right)}$$

$$A_m = \frac{2\pi Lc}{\ln\left(\frac{(R + c)}{R}\right)} \quad (18)$$

For coolant

$$q_s = 2\pi(R + c)Lh(T_m - T_s)$$

$$A_{R+c} = 2\pi L(R + c) \quad (19)$$

Rearrange the equation 17, 18 and 19



Effect of Cladding and Coolant

$$T_m - T_s = \frac{q_s}{4\pi K_f L} = \frac{q''' R^2}{4k_f} \quad (20)$$

$$T_s - T_c = \frac{q_s \ln\left(\frac{R+c}{R}\right)}{2\pi K_c L} = \frac{q''' R^2 \ln\left(\frac{R+c}{R}\right)}{2k_c} \quad (21)$$

$$T_c - T_f = \frac{q_s}{2\pi(R+c)Lh} = \frac{q''' R^2}{2(R+c)h} \quad (22)$$

Add equations 20, 21 and 22

$$T_m - T_f = \frac{q''' R^2}{4k_f} + \frac{q''' R^2}{2} \left(\frac{\ln\left(\frac{R+c}{R}\right)}{k_c} + \frac{1}{(R+c)h} \right) \quad (23)$$



Effect of Cladding and Coolant

Or could be written,

$$T_m - T_f = \frac{q_s}{4\pi K_f L} + \frac{q_s \ln\left(\frac{R+c}{R}\right)}{2\pi K_c L} + \frac{q_s}{2\pi(R+c)Lh}$$

$$T_m - T_f = q_s \left(\frac{R}{2K_f A_R} + \frac{c}{K_c A_m} + \frac{1}{hA_{R+c}} \right)$$

$$q_s = \frac{T_m - T_f}{\left(\frac{R}{2K_f A_R} + \frac{c}{K_c A_m} + \frac{1}{hA_{R+c}} \right)} \quad (24)$$



Effect of Cladding and Coolant

$$q_s = \frac{T_m - T_f}{\left(\frac{R}{2K_f A_R} + \frac{c}{K_c A_m} + \frac{1}{h A_{R+c}}\right)} \quad (24)$$

where,

$$A_R = 2\pi LR$$

$$A_m = \frac{2\pi Lc}{\ln\left(\frac{R+c}{R}\right)}$$

$$A_{R+c} = 2\pi L(R+c)$$

In the usual case, where c is very small compared to R , A_m may be taken, with little error, to be the arithmetic mean area of the cladding

$$A_m = \frac{(A_R + A_{R+c})}{2} = [2\pi R + 2\pi(R+c)] \frac{L}{2}$$



Example 1

A fuel rods 0.158 in diameter, 30.5 in long. The fuel rods are 26 w/o enriched uranium clad in 0.005-in stainless steel. The effective thermal neutron flux at the point of maximum temperature (slightly above the centre of the core) is 5.4×10^{13} *neutrons/sec · cm²*. The temperature at the center of the fuel rod where q'' is the largest, is 1,220 °F. Calculate the temperatures at the fuel-cladding interface at the outer surface of the cladding. The effective fission cross section σ_f is 364 b.

$D = 0.158 \text{ in}$	$\phi = 5.4 \times 10^{13} \text{ n/see cm}^2$
$c = 0.005 \text{ in}$	$T_m = 1220 \text{ }^\circ\text{F}$
$L = 30.5 \text{ in}$	$\sigma_f = 364 \text{ b}$
$\rho = 18.3 \text{ g/cm}^3$	$k_f \approx 20 \text{ Btu/hr ft }^\circ\text{F}$
	$k_c \approx 11 \text{ Btu/hr ft }^\circ\text{F}$



Example 1 (Solution)

$$q''' = GN_f \sigma_f \phi$$

$$N_f = \frac{0.60225 \times 10^{24}}{235.044} \times 18.3 \times 0.26 = 1.219 \times 10^{22}$$

$$q''' = GN_f \sigma_f \phi = 180 \times (1.219 \times 10^{22}) \times (364 \times 10^{-24}) \times (5.4 \times 10^{13})$$

$$q''' = 4.313 \times 10^{16} \times 1.55 \times 10^{-08}$$

$$q''' = 6.676 \times 10^{08} \text{ Btu/hr ft}^3$$



Example 1 (Solution)

$$T_m - T_s = \frac{q''' R^2}{4k_f}$$

$$T_s = T_m - \frac{q''' R^2}{4k_f}$$

$$T_s = 1220 - \frac{6.676 \times 10^{08} \left(\frac{0.158}{2 \times 12} \right)^2}{4 \times 20}$$

$$T_s = 1220 - \frac{3.709 \times 10^{08} \left(\frac{0.158}{2 \times 12} \right)^2}{2 \times 21.20} = 858^\circ F$$

$$T_s - T_c = \frac{q_s \ln|(R+c)/R|}{2 \pi k_C L} = \frac{q''' R^2 \ln|(R+c)/R|}{2 k_c}$$

$$T_c = T_s - \frac{q''' R^2 \ln\left(\frac{R+c}{R}\right)}{2 k_c}$$

$$T_c = 858 - \frac{3.709 \times 10^{08} \times \left(\frac{0.158}{2 \times 12} \right)^2 \times \ln\left(\frac{0.079+0.005}{0.079}\right)}{2 \times 11}$$

$$T_c = 858 - 44.84 = 778^\circ F$$



Heat Flow Out of Spherically Shaped Fuel

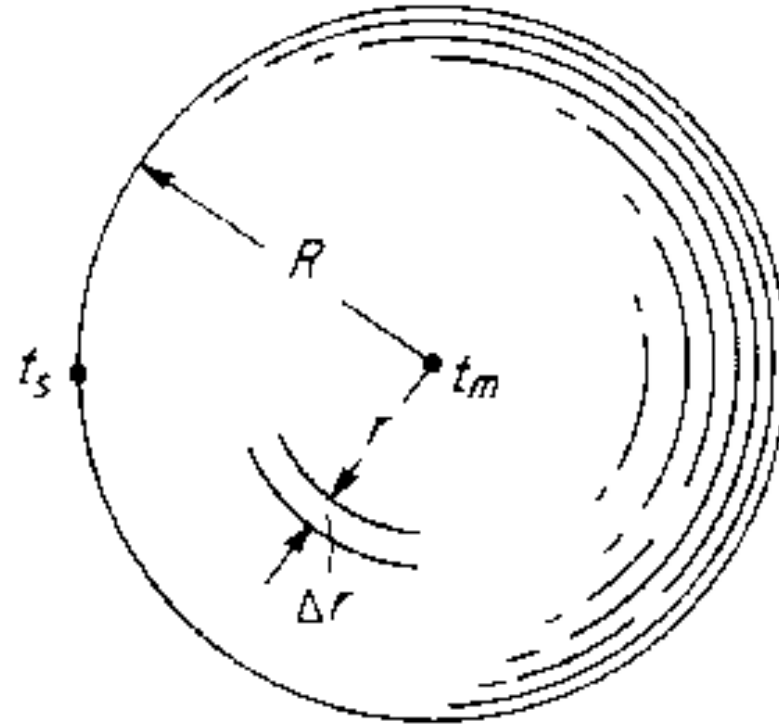
Poisson equation in spherical coordinates, also obtained by combining equation (25) and (26) to give

$$\nabla^2 T + \frac{q'''}{k} = 0 \quad (25)$$

$$\nabla^2 T = \left(\frac{d^2 T}{dr^2} + r^2 \frac{dT}{dr} \right)$$

$$\frac{d^2 T}{dr^2} + r^2 \frac{dT}{dr} + \frac{q'''}{k_f} = 0$$

$$\frac{1}{r^2} \frac{d}{dr} \left(r^2 \frac{dT}{dr} \right) + \frac{q'''}{k_f} = 0 \quad (26)$$





Heat Flow Out of Spherically Shaped Fuel

Integration of this equation for the boundary conditions $T(r)$ at r ,

B. C (1)

$$T = T_m \quad \text{at } r = 0$$

B. C (2)

$$T = T_s \quad \text{at } r = R$$

$$T(r) = T_m - \frac{q''' r^2}{6k_f} \quad (27)$$

$$T_m - T_s = \frac{q''' R^2}{6k_f} \quad (28)$$

The total heat generated, q_s , is given by the expression

$$q_s = \frac{4}{3} \pi R^3 q''' \quad (29)$$



Heat Flow Out of Spherically Shaped Fuel

Combining equation (28) and (29)

$$q_s = 8\pi R k_f (T_m - T_s) \quad (30)$$

Putting $4\pi R^2 = A_s$ the total peripheral area of the spherical element, and rearranging give

$$q_s = 2k_f A_s \left(\frac{T_m - T_s}{R} \right) \quad (31)$$



Thank You

Stay safe!